Analysis of Multipactor Effect in a Partially Dielectric-Loaded Rectangular Waveguide

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Abstract-This paper presents a study of the multipactor effect in a partially dielectric-loaded rectangular waveguide. 2 To obtain the simulations presented in this paper, a detailed 3 analysis of the dynamics of the electron inside this waveguide has been performed, taking into account the radio frequency 5 electromagnetic fields propagating in the waveguide and the dc 6 electric field that appears because of the charging of the dielectric layer. This electrostatic field is obtained by computing the electric potential produced by an arbitrary charge distribution on the 9 10 dielectric layer in a dielectric-loaded waveguide. The electron trajectory is then found by numerically solving the equations of 11 motion. The results obtained show that multipactor discharges 12 do turn off by themselves under certain circumstances when they 13 occur in such dielectric-loaded waveguide. 14

Index Terms-Multipactor, secondary emission, waveguide.

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I. INTRODUCTION

ULTIPACTOR is a high-power resonant electron dis-17 charge frequently observed in the microwave and 18 millimeter-wave subsystems operating under vacuum condi-19 tions [1] present in a wide range of different scenarios, such 20 as passive components of satellite communication payloads, 21 traveling-wave, tubes or particle accelerators. In an ultrahigh 22 vacuum environment, the free electrons inside a microwave 23 device are accelerated by the radio frequency (RF) electro-24 magnetic fields, impacting against its metallic walls. If the 25 electron impact energy is high enough, one or more secondary 26 electrons might be released from the surface. When some 27 resonance conditions are satisfied, the secondary electrons 28 get synchronized with the RF fields, and the electron pop-29 ulation inside the device grows exponentially leading to a 30 multipactor discharge. This multipactor discharge has some 31 negative effects that degrade the device performance: increase 32

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of signal noise and reflected power, heating of the device walls, outgassing, detuning of resonant cavities, and even the partial or total destruction of the component.

Multipactor research lines are aimed to study and characterize the phenomenon to predict the conditions for its 37 appearance [2], [3]. Some RF devices, such as filters, multiplexers, and RF satellite payloads, include dielectric materials commonly employed as resonators and supporting elements. In [4], it is presented a review of multipactor discharge on metals and dielectric windows that takes into account the surface materials, and the effects of space charge and cavity 43 loading. The multipactor effect including the presence of 44 dielectric materials in single-surface multipactor regime has 45 been widely investigated in the context of particle accelerators; for instance, in ceramic RF windows [5], [6] and in aluminabased dielectric-loaded accelerating structures [7]. In contrast, very few contributions can be found about multipactor breakdown on dielectrics in the scenario of RF systems for 50 space applications [8]–[10], and mostly under the parallel-51 plate waveguide approximation. In [11] and [12], the effective 52 electron model (EEM) has been successfully used for simulations of multipaction experiments in coaxial transmission lines considering the presence of external magnetic static fields, demonstrating the validity of this method in com-56 plex scenarios. The multipactor inside an empty rectangular 57 waveguide has also been studied in [13] and [14], where the 58 conventional resonance theory gives correct predictions for the multipactor threshold if the height of the waveguide is very small and first-order resonance multipactor dominates. When 61 the waveguide height exceeds a certain critical value, which depends on the waveguide width, an accurate prediction of the multipactor threshold requires considering the RF fields inside 64 the waveguide without approximations. Therefore, there is a 65 need to accurately predict the electron discharge on devices 66 involving partially dielectric-loaded rectangular waveguides, which are of more practical interest for satellite technology. 68 The main aim of this investigation is to extend the results of previous works [8]-[10], where an EEM was successfully applied to study the multipactor in a parallel-plate dielectric-71 loaded waveguide, to the analysis of multipactor effect in a partially dielectric-loaded rectangular waveguide.

In Section II, the theoretical model employed for the simulations is discussed. In Section III, the multipactor prediction results of an empty rectangular waveguide are analyzed and compared with results from the technical literature for validation purposes. Then, the susceptibility chart of a partially

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Fig. 1. Geometry and dimensions of the problem under investigation.

⁷⁹ dielectric-loaded rectangular waveguide is obtained with the
⁸⁰ developed model, and the time evolution of a discharge in
⁸¹ this waveguide is studied and discussed. Finally, Section IV
⁸² outlines the main conclusions of this paper.

II. THEORY

84 A. Computation of RF and DC Fields in a Partially

85 Dielectric-Loaded Waveguide

Fig. 1 shows the scheme of a partially dielectric-loaded 86 rectangular waveguide of width a and height b, and whose 87 dielectric material has relative permittivity ϵ_r . In the problem under study, the dielectric slab of thickness h and width a is placed over the bottom waveguide wall, being d the empty 90 waveguide height where the effective electron travels (see 91 Fig. 1). The RF electromagnetic field is assumed to propagate 92 along the positive direction of the z-axis. For the sake of 93 simplicity, the waveguide is supposed to be infinite along the 94 z direction, and a time-harmonic dependence of the type $e^{j\omega t}$ 95 is implicitly assumed, with $f = \omega/2\pi$ being the frequency and t the time measured in the laboratory reference sys-97 tem. To analyze the multipactor evolution in this waveguide, 98 a multipactor simulation code based on the Monte Carlo 99 method has been developed. The software code, similar to the 100 one described in [8] and [9], employs the single EEM [15]. 101 This assumption avoids the consideration of space-charge 102 effects, what is a strong simplification. The space-charge 103 effects are often neglected in the analysis of the first stages of 104 the multipactor discharge [13], [14], but they are doubtless 105 important at high electron populations when the discharge 106 is fully developed. Simulation results of some published 107 works [16], [17] indicate an important role of space-charge in 108 the evolution of the multipactor process to a saturation stage. 109 In this paper, however, we are mainly interested in studying 110 the influence of dielectric charging in the multipactor process. 111 The inclusion of space-charge effects, although providing a 112 more realistic description of the global process, would increase 113 the computational burden very much, as the dc field due 114 to dielectric charging has to be evaluated in every effective 115 electron position. In addition, the interpretation of simulation 116 results would become difficult, as dielectric charging and space 117 charge can both lead to a repulsion of the freshly emitted 118 secondary electrons back to the surface. 119

The effective electron at $\mathbf{r} = (x, y, z)$ can move in the air region of height *d* of the rectangular waveguide. The electromagnetic fields \mathbf{E}_{RF} and \mathbf{H}_{RF} acting on the effective electron correspond to the modes of the partially dielectricloaded rectangular waveguide (Fig. 1), which are hybrid modes of TM^y and TE^y kinds [18]. We have restricted our study to the monomode regime, where only the fundamental mode, TM_{10}^y , ¹²⁶ propagates in the waveguide. The instantaneous field vectors ¹²⁷ interacting with the effective electron are given by ¹²⁸

$$\mathbf{E}_{\rm RF}(x, y, z, t) = E_0 \Re(\mathbf{e}(x, y) e^{j(\omega t - \beta z + \varphi_0)})$$
(1a) 129

$$\mathbf{H}_{\rm RF}(x, y, z, t) = H_0 \Re(\mathbf{h}(x, y) e^{j(\omega t - \beta z + \varphi_0)})$$
(1b) 130

where φ_0 is the initial phase and E_0 , H_0 are the constants related to the transmitted power in the waveguide. The modal fields $\mathbf{e}(x, y)$ and $\mathbf{h}(x, y)$ and the propagation constant β of the TM^y₁₀ mode can be found in [18] and [19]. These expressions can be directly extended if higher order modes must be taken into account (i.e., in waveguide discontinuities) by using the mode-matching technique.

The key to understanding the mechanism of a multipactor 138 discharge is to study the behavior of the electrons within 139 the waveguide, which are accelerated by the aforementioned 140 electromagnetic fields E_{RF} and H_{RF} . In this way, sooner or 141 later, these fields will make an electron impact with any surface 142 of the rectangular waveguide, which can result in the emission 143 or absorption of secondary electrons. If the impacts occur on 144 the dielectric surface, unlike the case of impacts on the metallic 145 walls, the secondary electrons emitted by the dielectric give 146 rise to positive charges at the impact positions on the dielectric 147 surface, while the electrons absorbed in the dielectric layer 148 will generate negative charges in it. These charges, which are 149 located on the dielectric surface at positions $\mathbf{r}' = (x', 0, z')$, 150 give rise to an electrostatic field E_{dc} , which has to be added to 151 the RF fields to obtain accurately the trajectory of the electrons 152 inside the waveguide. In order to determine the electrostatic 153 field, $\mathbf{E}_{dc}(x, y, z) = -\nabla \phi(x, y, z)$, generated by the charges 154 on the dielectric, the potential $\phi(x, y, z)$ inside the waveguide 155 has to be first calculated. Using the superposition, the potential 156 in the waveguide due to the set of charges Q_i on the dielectric 157 surface can be obtained by adding the individual contribution 158 of each charge 159

$$\phi(x, y, z) = \sum_{i} G(x - x'_{i}, y, |z - z'_{i}|) Q_{i}(x'_{i}, 0, z'_{i}) \quad (2) \quad {}_{160}$$

where G(x, y, z) is the electrostatic potential due to a unit point charge, that is, Green's function for this problem. 162

The above-mentioned Green's function, G(x, y, z), is the solution to the following Laplace's equation [20], [21]: 164

$$\nabla \cdot [\epsilon_r(y)\nabla G(x, y, z)] = -\frac{1}{\epsilon_0}\delta(x - x')\delta(y)\delta(z) \qquad (3) \quad {}_{165}$$

where ϵ_0 is the free-space dielectric permittivity and the position of the unit charge is taken at (x', 0, 0) for convenience. Both the geometric characteristics and the linear nature of the problem under consideration makes that the Dirac delta functions can be expressed as [21] 170

$$\delta(x - x') = \frac{2}{a} \sum_{n=1}^{\infty} \sin(k_{xn}x) \sin(k_{xn}x')$$
(4) (4)

$$\delta(z) = \frac{1}{2\pi} \int_{-\infty}^{\infty} e^{-jk_z z} dk_z \tag{5}$$

where $k_{xn} = n\pi/a$ and k_z is the spectral Fourier variable along the longitudinal direction *z*. The above-mentioned expressions 174

come from the fact that the eigenfunctions of the differential operator are sinusoidal functions along *x*-axis and complex exponential functions along the *z*-axis, respectively. This is equivalent to apply the discrete sine transform along the *x*-axis and the integral transform along the *z*-axis, namely,

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$$G = \frac{1}{\pi a} \int_{-\infty}^{\infty} dk_z \, e^{-jk_z z} \sum_{n=1}^{\infty} \sin(k_{xn}x) \sin(k_{xn}x') \, \widetilde{G} \quad (6)$$

¹⁸¹
$$\widetilde{G} = \int_{-\infty}^{\infty} dz \, e^{jk_z z} \sum_{n=1}^{\infty} \sin(k_{xn}x) \sin(k_{xn}x') \, G \tag{7}$$

where G = G(x, x', y, z) and $\widetilde{G} = \widetilde{G}(k_{xn}, k_z; y)$.

According to the above-mentioned considerations, (3) can be expressed as the following ordinary differential equation for the spectral Green's function \tilde{G} :

$$\left\{\frac{d}{dy}\epsilon_r(y)\frac{d}{dy} - k_t^2\right\}\widetilde{G} = -\frac{\delta(y)}{\epsilon_0}$$
(8a)

$$G(y = -h) = 0 \tag{8b}$$

$$G(y=d) = 0 \tag{8c}$$

where $k_t^2 = k_{xn}^2 + k_z^2$. Solving (8), the following expression for \widetilde{G} is obtained in the air region $y \ge 0$:

¹⁹¹
$$\widetilde{G}(k_{xn}, k_z; y) = \frac{\sinh[k_t(d-y)]}{\epsilon_0 k_t [\epsilon_r \coth(k_t h) + \coth(k_t d)] \sinh(k_t d)}.$$
¹⁹²(9)

Green's function in the spatial domain, G, is achieved by replacing (9) into (6) to give

¹⁹⁵
$$G(x, x', y, z)$$
¹⁹⁶
$$= \frac{2}{\epsilon_0 \pi a} \sum_{n=1}^{\infty} \sin(k_{xn}x) \sin(k_{xn}x')$$
¹⁹⁷
$$\times \int_0^{\infty} \frac{\sinh[k_t(d-y)]\cos(k_z z)}{k_t[\epsilon_r \coth(k_t h) + \coth(k_t d)]\sinh(k_t d)} dk_z.$$
 (10)

¹⁹⁸ In (10), if the point charge is placed at $z' \neq 0$, z must ¹⁹⁹ be replaced by (z - z'). Here, it is worth noting that very ²⁰⁰ efficient numerical summation and integration techniques have ²⁰¹ to be employed to compute Green's function with sufficient ²⁰² accuracy and tolerable CPU times [22].

Once Green's function has been calculated, the E_{dc} field is obtained by numerical differentiation of (2) by means of the central difference technique.

B. Multipactor Evolution in the Partially Dielectric-Loaded Waveguide

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Once the RF and dc fields are known at any instant t, the electron dynamics inside the waveguide can be computed, which is governed by the Lorentz force and related to its linear momentum

$$\mathbf{F}_L = q(\mathbf{E} + \mathbf{v} \times \mathbf{B}) = \frac{d\mathbf{p}}{dt}$$
(11)

where q = -e is the electron charge, **E** and **B** = μ_0 **H** are the total electric and magnetic fields (both RF and dc contributions) interacting with the electron, μ_0 is the freespace magnetic permeability, and **v** is the velocity vector of the electron. The linear relativistic momentum is defined as

$$\mathbf{p} = m_0 \gamma \, \mathbf{v} \tag{12}$$

where m_0 is the electron rest mass, $\gamma = 1/\sqrt{1 - (v/c)^2}$ the 219 Lorentz factor, v is the magnitude of the velocity vector, and 220 $c = 1/\sqrt{\mu_0 \epsilon_0}$ the speed of light in vacuum. Although the 221 relativistic correction in this equation can be discarded for the 222 typical power ranges of most space waveguide devices, it must 223 be considered when high velocities are reached $(v/c \ge 0.1)$, 224 as in high-power multipactor simulations. Expanding (11), 225 the following differential equation is obtained: 226

$$-\mathbf{E} - \mathbf{v} \times \mathbf{B} = M\gamma \,\mathbf{a} + \frac{M}{c^2}\gamma^3 (\mathbf{v} \cdot \mathbf{a})\mathbf{v}$$
(13) 22

where **a** is the acceleration vector and $M = m_0/e$. The ²²⁸ differential equation to be solved becomes ²²⁹

$$\ddot{\mathbf{r}} = \frac{-\dot{\mathbf{r}} \times \mathbf{B} - \mathbf{E} + \dot{\mathbf{r}} \cdot (\dot{\mathbf{r}} \cdot \mathbf{E})/c^2}{M\gamma}.$$
(14) 230

The electron trajectory is found by numerically solving 231 the above-mentioned equations of motion. For that purpose, 232 a Velocity Verlet algorithm [23] has been used, which assures 233 sufficient accuracy and good efficiency provided the time step 234 is small enough. Regarding this last point, in order to improve 235 the accuracy and efficiency of the simulation, the following 236 adaptive time step has been applied in the proximity of the 237 waveguide walls, depending on the electron position: 238

$$\Delta t = \frac{\Delta t_0}{1 + \xi \left(\frac{x - a/2}{a/2}\right)^2 + \xi \left(\frac{y - d/2}{d/2}\right)^2}$$
(15) 236

where Δt_0 is the initial reference time step, ξ is a constant value (in this case, a value of 4.0 has been chosen), and x and y are the coordinates of the electron position. 240

As mentioned above, the computed electrons trajectories 243 may lead to an eventual impact with a surface. Each collision 244 can result in the emission or absorption of secondary electrons. 245 A relevant growth in the electron density can develop if the 246 electrons hit the walls with the appropriate energy and at 247 suitable instants. The number of electrons emitted or absorbed 248 after each impact is determined by the value of the Secondary 249 Electron Yield (SEY) parameter δ (δ > 1 if secondary 250 electrons are emitted, and $\delta < 1$ if they are absorbed). The 251 SEY is modeled by a modification of Vaughan's model [24] 252 that includes the effect of reflected electrons for low impact 253 energies of primary electrons, which has to be accounted for to 254 obtain accurate results [25], [26] in agreement with experimen-255 tal data obtained in [27], [28]. The SEY properties for surface 256 materials can be defined by the following parameters: the 257 primary electron impact kinetic energies which yield $\delta = 1$, 258 W_1 , and W_2 ; the impact energy W_{max} necessary for a primary 259 electron to yield $\delta = \delta_{max}$, which is the maximum value of the 260 SEY function; and the value of the primary electron impact 261 energy $W_0(\delta = 0)$ that limits the region of elastic collisions. 262

When a multipactor discharge evolves in the partially dielectric-loaded waveguide under study, the dc field distribution has to be updated after each electron impacts on the 265

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Material	W _{max} (eV)	W ₁ (eV)	$W_2(eV)$	W ₀ (eV)	δ_{max}
Niobium	200.0	33	1500	-	1.6
Silver	165.0	30	5000	15.99	2.22
Teflon	271.7	27	5000	6.81	2.47

TABLE I Secondary Electron Emission Yield Properties of Different Materials [31], [32]

dielectric surface. However, tracking the evolution of all the 266 electrons involved in the multipactor discharge would suppose 267 a big computational cost. Thus, we have made use of an EEM, 268 but considering both the spread in secondary emission energy 269 and the angle of the secondary electrons after each impact on 270 the waveguide walls. This assumption has proved to account 271 properly for the charging of the dielectric material, given that 272 the discharging time for dielectrics is much higher than the 273 typical time for a multipactor discharge. Thus, in the EEM 274 assumed in this paper, after the effective electron impacts at 275 time t with any surface, $N_i(t)$ is modified according to the δ 276 value provided by the SEY function as follows: 277

$$N_i(t + \Delta t) = \delta N_i(t) \tag{16}$$

where $N_i(t)$ represents the population of the ee inside the waveguide at the instant *t*, and Δt is the time step used in the simulations.

The secondary electron departure kinetic energy E_s after each electron impact is assumed to fit the following probability density function [29]:

$$\frac{dp(E_s)}{dE_s} = C \exp\left[-\frac{\ln^2(E_s/E_m)}{2\tau^2}\right]$$
(17)

where *C* is a normalization constant, the parameter τ (typical values 0.7–0.8) determines the width of the distribution and *E_m* (typical values 3–4 eV) is the energy of the maximum of the spectrum. Finally, the secondary electrons after inelastic impacts are emitted following a cosine distribution of the polar angle.

III. NUMERICAL RESULTS AND DISCUSSION

An in-house simulation computer-aided design (CAD) tool 293 based on the Monte Carlo method described in Section II has 294 been developed to analyze the multipactor effect in partially 295 dielectric-loaded rectangular waveguides. The first problem 296 analyzed consists of an empty rectangular waveguide previ-297 ously studied in [14], whose multipactor prediction results 298 have been used for validation purposes. The rectangular 299 waveguide has dimensions a = 43.2 cm and b = 10.2 cm, 300 and is excited by a time-harmonic signal at f = 500 MHz. 301 The material of the waveguide walls is niobium, whose SEY 302 properties are given in Table I and can be expressed with 303 the simple model proposed in [30]. In the algorithm of the 304 simulator used in [14], for each RF power considered in the 305 waveguide, the initial electron is launched at x = a/2, and 306 the simulation is run 42 times, corresponding to 42 equidistant 307 phases of the RF field. The mean value of the final population 308 of electrons after 20 impacts of the ee against the walls is 309 calculated using all the 42 simulations. Also, the secondary 310



Fig. 2. Comparison with [14] of the mean value of N over all launch phases in a rectangular waveguide (a = 43.1 cm and b = 10.2 cm) driven at f = 500 MHz with a maximum of 20 impacts from a single initial launch location on the midline of the empty rectangular waveguide.

electrons generated after every collision are launched with an 311 energy of 2 eV normal to the impacting surface. The maximum 312 simulation lifetime of each ee is $t_{max} = 1000$ RF cycles, and 313 the simulation is stopped if the impact energy is lower than 314 0.1 eV or if the accumulated population of electrons is under 315 10^{-3} . To model the same simulation conditions, our CAD tool 316 has been adapted accordingly. In Fig. 2, the results of the mean 317 population of electrons, \overline{N} , computed with our code (black 318 lines) are compared with the curves presented in [14] (gray 319 lines). In this figure, we can see some high-risk multipactor 320 power regions. Both curves show a good agreement in the 321 shape and location of these multipactor windows. 322

Once the model has been validated for an empty 323 waveguide, next we analyze the multipactor effect in a par-324 tially dielectric-loaded rectangular waveguide. The selected 325 waveguide configuration for the multipactor analysis is a 326 nonstandard silver-plated rectangular waveguide of width a =327 19.05 mm and heigh b = 0.4 mm, in which a thin dielec-328 tric layer has been placed over the bottom surface of the 329 waveguide. A realistic dielectric material has been chosen 330 as teflon (DuPont Teflon fluorinated ethylene propylene Flu-331 oroplastic Film Type), which is a dielectric film commonly 332 used in space applications, of thickness h = 0.025 mm and 333 $\epsilon_r = 2.1$; thus d = b - h = 0.375 mm. Standard values for 334 the SEY parameters of silver [31] are given in Table I, and 335 SEY parameters of teflon have been measured at the ESA-VSC 336 High Power Space Materials Laboratory (Valencia, Spain) 337 [32]. First, a study of the susceptibility chart of this waveguide 338 has been performed. Since it is a partially dielectric-loaded 339 rectangular waveguide, the factor $f \times d$ is plotted in the 340 horizontal axis of the susceptibility chart. In the vertical 341 axis, it is plotted an effective voltage, Veff, which has been 342 calculated numerically as the line integral of the E_{y} component 343 of the electric field (evaluated at the center of the waveguide 344 x = a/2 from $y_1 = 0$ to $y_2 = d$. To obtain this susceptibility 345 chart for each $V_{\rm eff}$ and $f \times d$ pair, the simulation is run 346 72 times, corresponding to 72 equidistant initial phases of 347 the RF field separated 5°. In each run, an initial single ee is 348 launched at x = a/2 and z = 0 and at a random position 349 y_0 in the y-axis between y = 0 and y = d. The initial 350 electron is launched with a departure kinetic energy given by 351 the probability density function shown in (17) and following 352



Fig. 3. Comparison of the susceptibility chart of a rectangular waveguide partially filled with teflon (black points) with that of its equivalent empty waveguide (gray points). Red: operating point corresponding to $f \times d = 3.13 \text{ GHz} \cdot \text{mm}$ and $V_{\text{eff}} = 608 \text{ V}$.

a cosine distribution of the polar angle. Each simulation was 353 stopped after 100 RF cycles. In the empty waveguide, the 354 arithmetic mean of the final population of electrons after 355 100 RF cycles is calculated using all the 72 simulations. 356 If this mean value is greater than 1, then the multipactor 357 discharge is assumed to have occurred. However, in a partially 358 dielectric-loaded waveguide, it has been shown in previous 359 works [8]–[10] that the emission or absorption of electrons 360 by the dielectric surface gives rise to an increasing dc field 361 in the waveguide, which eventually turns off the discharge. 362 Thus, in this case, a minimum mean value of the magnitude 363 of E_{dc} field in the waveguide after 100 RF cycles is used as the 364 criterion to assume that a multipactor discharge has occurred 365 at a given operating point. 366

Fig. 3 shows the computed susceptibility chart of the rec-367 tangular waveguide partially filled with teflon (black points). 368 The lowest $f \times d$ value is above the cutoff frequency 369 of the fundamental mode in this waveguide. In this figure, 370 the susceptibility chart of the equivalent empty waveguide 371 with the same vertical air gap is also represented with gray 372 points for comparison. It can be checked that both the empty 373 and the partially dielectric-loaded waveguide with the same 374 vertical air gap show similar multipactor susceptibility charts, 375 given that the SEY properties of silver and teflon are similar. 376 This susceptibility chart is not generally applicable to any 377 rectangular waveguide with an air gap d, given that the 378 electromagnetic field distribution depends on the geometry 379 and dimensions of the dielectric layer with respect to the 380 waveguide dimensions, and also on its relative permittivity. 381

From the results shown earlier, and with the purpose of 382 having a better understanding of the dynamics of the electron 383 inside the partially dielectric-loaded waveguide, a point within 384 the multipactor region has been chosen (highlighted in red 385 in Fig. 3), corresponding to $V_{\rm eff} = 608$ V and $f \times d =$ 386 3.13 GHz · mm. In this case, the evolution of the multipactor 387 discharge in the partially dielectric-loaded waveguide under 388 study has been analyzed as a function of the time normalized 389 to the RF period. For this simulation, the electron is launched 390 with an initial phase of the RF field $\varphi_0 = 0^\circ$. Simulations 391 assuming different initial phases have been performed, and 392 similar results were obtained. The obtained simulation is 393 shown in Fig. 4, where it is plotted the y-coordinate followed 394 by the ee within the waveguide as a function of the normalized 395



Fig. 4. Trajectory (y-coordinate) of the ee in the air gap of the rectangular waveguide partially filled with teflon as a function of the RF cycle.



Fig. 5. Black line; time evolution of the total number of electrons N. Blue line; $E_{y,RF}$. Red line: $E_{y,dc}$ at the electron position.



Fig. 6. Distribution of normalized charges $\overline{Q}_i = Q_i/e$ appearing on the dielectric surface.

time. In the selected multipactor regime, which is inside the 396 multipactor region, the electron initially collides with the 397 top metallic and bottom dielectric surface consecutively in 398 what seems to be a first-order multipactor process during 399 the first 17 RF cycles, remaining in the vicinity of x =400 a/2 and z = 0—given that the electron has nearly no 401 acceleration in such directions. As shown in Fig. 5, in the 402 first cycles, the total number of electrons N (black solid 403 line) follows an exponential growth. This progressive growing 404 of N makes that the number of charges appearing on the 405 dielectric surface increases, the number which is proportional 406 to the emitted or absorbed electrons in each impact, as seen 407 in Fig. 6 (positive charges are represented with red circles, 408 while negative charges are represented with blue circles; the 409 circles' size is proportional to the charge magnitude in log 410 scale). Such charges on the dielectric interface give rise to the 411



Transverse distribution of the dc electric field in the proximity of Fig. 7. the main charge point in the waveguide air region at RF cycle 26.2, at z =0.45 mm.

appearance of an electrostatic field in the empty gap during 412 the time between impacts. Once the population of electrons 413 reaches a significant number ($N \approx 10^9$ in the conditions under 414 study), the y-component of the dc field, $E_{y,dc}$ [which has been 415 plotted in Fig. 5 with red line at the positions (x, y, z) where 416 the effective electron is located in the displayed instants in 417 this figure] becomes comparable to $E_{y,RF}$, and the effective 418 electron is unable to keep up with its previous multipactor 419 synchronization. From this moment on, the dc field makes that, 420 in some impacts, the electrons collide with the top metallic 421 bottom dielectric surface much sooner or later than the or 422 instants when the RF electric field changes its sign, which 423 implies low impact energy collisions so that electrons are 424 absorbed in such impacts. In collisions at the dielectric surface, 425 the absorption or emission of electrons yield the appearance 426 of growing charges on the dielectric layer, contributing to a 427 higher dc field acting on the waveguide. The distribution of 428 this high dc field in the proximity of the main charge point 429 in the waveguide air region is shown in Fig. 7 at RF cycle 430 26.2 in the plane z = 0.45 mm (corresponding to the z 431 position of the electron at this instant). The action of this field 432 may result in the appearance of a single-surface multipactor 433 regime in the dielectric surface [see the y position of the 434 electron in Fig. 4 (inset) from RF cycle 26.2], with successive 435 low impact energy collisions, which eventually leads to the 436 turning off of the discharge itself (as can be appreciated in 437 Fig. 5 from RF cycle 26.2 on). From this instant, the dc field 438 distribution in the waveguide remains nearly constant, given 439 that N drops very quickly. The final value of the y-component 440 of the dc field accounts for the balance between the emitted 441 and absorbed electrons by the dielectric surface in the whole 442 process. Then, although the final population of electrons after 443 RF cycle 100 is 0, the remaining high dc field in the waveguide 444 indicates that a multipactor discharge has taken place in the 445 waveguide in this simulation. It is worth mentioning that the 446 observed turning OFF of the discharge observed in the last 447 stages of the multipactor evolution in this waveguide has been 448 speeded up due to the use of the EEM, although this does not 449 change qualitatively the dynamics of the discharge under these 450 conditions. 451

IV. CONCLUSION

A study of the multipactor effect in a partially dielectric-453 loaded rectangular waveguide has been carried out. In this 454

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paper, we have considered the RF electromagnetic fields 455 (obtained with a very efficient vectorial modal method) as 456 well as the dc field caused by the appearance of a charge 457 distribution in the dielectric layer. The solution of the elec-458 trostatic problem has required the use of different numer-459 ical integration techniques and interpolation methods. The 460 electron trajectory has been numerically solved by using a 461 Velocity Verlet algorithm, providing sufficient accuracy and 462 good efficiency. As a first example, the multipactor prediction 463 results of an empty rectangular waveguide have been obtained 464 for validation purposes. Second, the susceptibility chart of 465 a partially dielectric-loaded rectangular waveguide has been 466 computed, and the time evolution of a discharge in this 467 waveguide has been studied and discussed. The performed 468 simulations reveal that multipactor discharges in this type of 469 dielectric-loaded waveguides turn OFF by themselves due to 470 the electrostatic field associated with the dielectric surface 471 charges that evolve with the multipactor process. 472

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Analysis of Multipactor Effect in a Partially Dielectric-Loaded Rectangular Waveguide

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Abstract-This paper presents a study of the multipactor effect in a partially dielectric-loaded rectangular waveguide. 2 To obtain the simulations presented in this paper, a detailed 3 analysis of the dynamics of the electron inside this waveguide has been performed, taking into account the radio frequency 5 electromagnetic fields propagating in the waveguide and the dc 6 electric field that appears because of the charging of the dielectric layer. This electrostatic field is obtained by computing the electric potential produced by an arbitrary charge distribution on the 9 10 dielectric layer in a dielectric-loaded waveguide. The electron trajectory is then found by numerically solving the equations of 11 motion. The results obtained show that multipactor discharges 12 do turn off by themselves under certain circumstances when they 13 occur in such dielectric-loaded waveguide. 14

Index Terms-Multipactor, secondary emission, waveguide.

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I. INTRODUCTION

ULTIPACTOR is a high-power resonant electron dis-17 charge frequently observed in the microwave and 18 millimeter-wave subsystems operating under vacuum condi-19 tions [1] present in a wide range of different scenarios, such 20 as passive components of satellite communication payloads, 21 traveling-wave, tubes or particle accelerators. In an ultrahigh 22 vacuum environment, the free electrons inside a microwave 23 device are accelerated by the radio frequency (RF) electro-24 magnetic fields, impacting against its metallic walls. If the 25 electron impact energy is high enough, one or more secondary 26 electrons might be released from the surface. When some 27 resonance conditions are satisfied, the secondary electrons 28 get synchronized with the RF fields, and the electron pop-29 ulation inside the device grows exponentially leading to a 30 multipactor discharge. This multipactor discharge has some 31 negative effects that degrade the device performance: increase 32

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of signal noise and reflected power, heating of the device walls, outgassing, detuning of resonant cavities, and even the partial or total destruction of the component.

Multipactor research lines are aimed to study and characterize the phenomenon to predict the conditions for its appearance [2], [3]. Some RF devices, such as filters, multiplexers, and RF satellite payloads, include dielectric materials commonly employed as resonators and supporting elements. In [4], it is presented a review of multipactor discharge on metals and dielectric windows that takes into account the surface materials, and the effects of space charge and cavity loading. The multipactor effect including the presence of dielectric materials in single-surface multipactor regime has been widely investigated in the context of particle accelerators; for instance, in ceramic RF windows [5], [6] and in aluminabased dielectric-loaded accelerating structures [7]. In contrast, very few contributions can be found about multipactor breakdown on dielectrics in the scenario of RF systems for space applications [8]–[10], and mostly under the parallelplate waveguide approximation. In [11] and [12], the effective electron model (EEM) has been successfully used for simulations of multipaction experiments in coaxial transmission lines considering the presence of external magnetic static fields, demonstrating the validity of this method in complex scenarios. The multipactor inside an empty rectangular waveguide has also been studied in [13] and [14], where the conventional resonance theory gives correct predictions for the multipactor threshold if the height of the waveguide is very small and first-order resonance multipactor dominates. When the waveguide height exceeds a certain critical value, which depends on the waveguide width, an accurate prediction of the multipactor threshold requires considering the RF fields inside the waveguide without approximations. Therefore, there is a need to accurately predict the electron discharge on devices involving partially dielectric-loaded rectangular waveguides, which are of more practical interest for satellite technology. The main aim of this investigation is to extend the results of previous works [8]-[10], where an EEM was successfully applied to study the multipactor in a parallel-plate dielectricloaded waveguide, to the analysis of multipactor effect in a partially dielectric-loaded rectangular waveguide.

In Section II, the theoretical model employed for the simulations is discussed. In Section III, the multipactor prediction results of an empty rectangular waveguide are analyzed and compared with results from the technical literature for validation purposes. Then, the susceptibility chart of a partially

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Fig. 1. Geometry and dimensions of the problem under investigation.

⁷⁹ dielectric-loaded rectangular waveguide is obtained with the
⁸⁰ developed model, and the time evolution of a discharge in
⁸¹ this waveguide is studied and discussed. Finally, Section IV
⁸² outlines the main conclusions of this paper.

II. THEORY

A. Computation of RF and DC Fields in a Partially

85 Dielectric-Loaded Waveguide

Fig. 1 shows the scheme of a partially dielectric-loaded 86 rectangular waveguide of width a and height b, and whose 87 dielectric material has relative permittivity ϵ_r . In the problem under study, the dielectric slab of thickness h and width a is placed over the bottom waveguide wall, being d the empty 90 waveguide height where the effective electron travels (see 91 Fig. 1). The RF electromagnetic field is assumed to propagate 92 along the positive direction of the z-axis. For the sake of 93 simplicity, the waveguide is supposed to be infinite along the 94 z direction, and a time-harmonic dependence of the type $e^{j\omega t}$ 95 is implicitly assumed, with $f = \omega/2\pi$ being the frequency and t the time measured in the laboratory reference sys-97 tem. To analyze the multipactor evolution in this waveguide, 98 a multipactor simulation code based on the Monte Carlo aa method has been developed. The software code, similar to the 100 one described in [8] and [9], employs the single EEM [15]. 101 This assumption avoids the consideration of space-charge 102 effects, what is a strong simplification. The space-charge 103 effects are often neglected in the analysis of the first stages of 104 the multipactor discharge [13], [14], but they are doubtless 105 important at high electron populations when the discharge 106 is fully developed. Simulation results of some published 107 works [16], [17] indicate an important role of space-charge in 108 the evolution of the multipactor process to a saturation stage. 109 In this paper, however, we are mainly interested in studying 110 the influence of dielectric charging in the multipactor process. 111 The inclusion of space-charge effects, although providing a 112 more realistic description of the global process, would increase 113 the computational burden very much, as the dc field due 114 to dielectric charging has to be evaluated in every effective 115 electron position. In addition, the interpretation of simulation 116 results would become difficult, as dielectric charging and space 117 charge can both lead to a repulsion of the freshly emitted 118 secondary electrons back to the surface. 119

The effective electron at $\mathbf{r} = (x, y, z)$ can move in the air region of height *d* of the rectangular waveguide. The electromagnetic fields \mathbf{E}_{RF} and \mathbf{H}_{RF} acting on the effective electron correspond to the modes of the partially dielectricloaded rectangular waveguide (Fig. 1), which are hybrid modes of TM^y and TE^y kinds [18]. We have restricted our study to the monomode regime, where only the fundamental mode, TM_{10}^y , ¹²⁶ propagates in the waveguide. The instantaneous field vectors ¹²⁷ interacting with the effective electron are given by ¹²⁸

$$\mathbf{E}_{\rm RF}(x, y, z, t) = E_0 \Re(\mathbf{e}(x, y) e^{j(\omega t - \beta z + \varphi_0)})$$
(1a) 129

$$\mathbf{H}_{\rm RF}(x, y, z, t) = H_0 \Re(\mathbf{h}(x, y) e^{j(\omega t - \beta z + \varphi_0)})$$
(1b) 130

where φ_0 is the initial phase and E_0 , H_0 are the constants related to the transmitted power in the waveguide. The modal fields $\mathbf{e}(x, y)$ and $\mathbf{h}(x, y)$ and the propagation constant β of the TM^y₁₀ mode can be found in [18] and [19]. These expressions can be directly extended if higher order modes must be taken into account (i.e., in waveguide discontinuities) by using the mode-matching technique.

The key to understanding the mechanism of a multipactor 138 discharge is to study the behavior of the electrons within 139 the waveguide, which are accelerated by the aforementioned 140 electromagnetic fields E_{RF} and H_{RF} . In this way, sooner or 141 later, these fields will make an electron impact with any surface 142 of the rectangular waveguide, which can result in the emission 143 or absorption of secondary electrons. If the impacts occur on 144 the dielectric surface, unlike the case of impacts on the metallic 145 walls, the secondary electrons emitted by the dielectric give 146 rise to positive charges at the impact positions on the dielectric 147 surface, while the electrons absorbed in the dielectric layer 148 will generate negative charges in it. These charges, which are 149 located on the dielectric surface at positions $\mathbf{r}' = (x', 0, z')$, 150 give rise to an electrostatic field E_{dc} , which has to be added to 151 the RF fields to obtain accurately the trajectory of the electrons 152 inside the waveguide. In order to determine the electrostatic 153 field, $\mathbf{E}_{dc}(x, y, z) = -\nabla \phi(x, y, z)$, generated by the charges 154 on the dielectric, the potential $\phi(x, y, z)$ inside the waveguide 155 has to be first calculated. Using the superposition, the potential 156 in the waveguide due to the set of charges Q_i on the dielectric 157 surface can be obtained by adding the individual contribution 158 of each charge 159

$$\phi(x, y, z) = \sum_{i} G(x - x'_{i}, y, |z - z'_{i}|) Q_{i}(x'_{i}, 0, z'_{i}) \quad (2) \quad 100$$

where G(x, y, z) is the electrostatic potential due to a unit point charge, that is, Green's function for this problem. 162

The above-mentioned Green's function, G(x, y, z), is the solution to the following Laplace's equation [20], [21]: 164

$$\nabla \cdot [\epsilon_r(y)\nabla G(x, y, z)] = -\frac{1}{\epsilon_0}\delta(x - x')\delta(y)\delta(z) \qquad (3) \quad {}_{165}$$

where ϵ_0 is the free-space dielectric permittivity and the position of the unit charge is taken at (x', 0, 0) for convenience. Both the geometric characteristics and the linear nature of the problem under consideration makes that the Dirac delta functions can be expressed as [21] 170

$$\delta(x - x') = \frac{2}{a} \sum_{n=1}^{\infty} \sin(k_{xn}x) \sin(k_{xn}x')$$
(4) (4)

$$\delta(z) = \frac{1}{2\pi} \int_{-\infty}^{\infty} e^{-jk_z z} dk_z \tag{5}$$

where $k_{xn} = n\pi/a$ and k_z is the spectral Fourier variable along the longitudinal direction *z*. The above-mentioned expressions 174

come from the fact that the eigenfunctions of the differential operator are sinusoidal functions along *x*-axis and complex exponential functions along the *z*-axis, respectively. This is equivalent to apply the discrete sine transform along the *x*-axis and the integral transform along the *z*-axis, namely,

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$$G = \frac{1}{\pi a} \int_{-\infty}^{\infty} dk_z \, e^{-jk_z z} \sum_{n=1}^{\infty} \sin(k_{xn}x) \sin(k_{xn}x') \, \widetilde{G} \quad (6)$$

$$\widetilde{G} = \int_{-\infty}^{\infty} dz \, e^{jk_z z} \sum_{n=1}^{\infty} \sin(k_{xn}x) \sin(k_{xn}x') \, G \tag{7}$$

where G = G(x, x', y, z) and $\widetilde{G} = \widetilde{G}(k_{xn}, k_z; y)$.

According to the above-mentioned considerations, (3) can be expressed as the following ordinary differential equation for the spectral Green's function \tilde{G} :

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$$\left\{\frac{d}{dy}\epsilon_r(y)\frac{d}{dy} - k_t^2\right\}\widetilde{G} = -\frac{\delta(y)}{\epsilon_0}$$
(8a)

$$G(y = -h) = 0$$
 (8b)

$$G(y=d) = 0 \tag{8c}$$

where $k_t^2 = k_{xn}^2 + k_z^2$. Solving (8), the following expression for \widetilde{G} is obtained in the air region $y \ge 0$:

¹⁹¹
$$\widetilde{G}(k_{xn}, k_z; y) = \frac{\sinh[k_t(d-y)]}{\epsilon_0 k_t[\epsilon_r \coth(k_t h) + \coth(k_t d)]\sinh(k_t d)}.$$
¹⁹² (9)

Green's function in the spatial domain, G, is achieved by replacing (9) into (6) to give

¹⁹⁵
$$G(x, x', y, z)$$
¹⁹⁶
$$= \frac{2}{\epsilon_0 \pi a} \sum_{n=1}^{\infty} \sin(k_{xn}x) \sin(k_{xn}x')$$
¹⁹⁷
$$\times \int_0^{\infty} \frac{\sinh[k_t(d-y)]\cos(k_z z)}{k_t[\epsilon_r \coth(k_t h) + \coth(k_t d)]\sinh(k_t d)} dk_z.$$
 (10)

¹⁹⁸ In (10), if the point charge is placed at $z' \neq 0$, z must ¹⁹⁹ be replaced by (z - z'). Here, it is worth noting that very ²⁰⁰ efficient numerical summation and integration techniques have ²⁰¹ to be employed to compute Green's function with sufficient ²⁰² accuracy and tolerable CPU times [22].

Once Green's function has been calculated, the E_{dc} field is obtained by numerical differentiation of (2) by means of the central difference technique.

B. Multipactor Evolution in the Partially Dielectric-Loaded Waveguide

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Once the RF and dc fields are known at any instant t, the electron dynamics inside the waveguide can be computed, which is governed by the Lorentz force and related to its linear momentum

$$\mathbf{F}_L = q(\mathbf{E} + \mathbf{v} \times \mathbf{B}) = \frac{d\mathbf{p}}{dt}$$
(11)

where q = -e is the electron charge, **E** and **B** = μ_0 **H** are the total electric and magnetic fields (both RF and dc contributions) interacting with the electron, μ_0 is the freespace magnetic permeability, and **v** is the velocity vector of the electron. The linear relativistic momentum is defined as

$$\mathbf{p} = m_0 \gamma \, \mathbf{v} \tag{12}$$

where m_0 is the electron rest mass, $\gamma = 1/\sqrt{1 - (v/c)^2}$ the 219 Lorentz factor, v is the magnitude of the velocity vector, and 220 $c = 1/\sqrt{\mu_0\epsilon_0}$ the speed of light in vacuum. Although the 221 relativistic correction in this equation can be discarded for the 222 typical power ranges of most space waveguide devices, it must 223 be considered when high velocities are reached $(v/c \ge 0.1)$, 224 as in high-power multipactor simulations. Expanding (11), 225 the following differential equation is obtained: 226

$$-\mathbf{E} - \mathbf{v} \times \mathbf{B} = M\gamma \,\mathbf{a} + \frac{M}{c^2}\gamma^3 (\mathbf{v} \cdot \mathbf{a})\mathbf{v}$$
(13) 22

where **a** is the acceleration vector and $M = m_0/e$. The ²²⁸ differential equation to be solved becomes ²²⁹

$$\ddot{\mathbf{r}} = \frac{-\dot{\mathbf{r}} \times \mathbf{B} - \mathbf{E} + \dot{\mathbf{r}} \cdot (\dot{\mathbf{r}} \cdot \mathbf{E})/c^2}{M\gamma}.$$
(14) 230

The electron trajectory is found by numerically solving 231 the above-mentioned equations of motion. For that purpose, 232 a Velocity Verlet algorithm [23] has been used, which assures 233 sufficient accuracy and good efficiency provided the time step 234 is small enough. Regarding this last point, in order to improve 235 the accuracy and efficiency of the simulation, the following 236 adaptive time step has been applied in the proximity of the 237 waveguide walls, depending on the electron position: 238

$$\Delta t = \frac{\Delta t_0}{1 + \xi \left(\frac{x - a/2}{a/2}\right)^2 + \xi \left(\frac{y - d/2}{d/2}\right)^2}$$
(15) 233

where Δt_0 is the initial reference time step, ξ is a constant value (in this case, a value of 4.0 has been chosen), and x and y are the coordinates of the electron position. 240

As mentioned above, the computed electrons trajectories 243 may lead to an eventual impact with a surface. Each collision 244 can result in the emission or absorption of secondary electrons. 245 A relevant growth in the electron density can develop if the 246 electrons hit the walls with the appropriate energy and at 247 suitable instants. The number of electrons emitted or absorbed 248 after each impact is determined by the value of the Secondary 249 Electron Yield (SEY) parameter δ (δ > 1 if secondary 250 electrons are emitted, and $\delta < 1$ if they are absorbed). The 251 SEY is modeled by a modification of Vaughan's model [24] 252 that includes the effect of reflected electrons for low impact 253 energies of primary electrons, which has to be accounted for to 254 obtain accurate results [25], [26] in agreement with experimen-255 tal data obtained in [27], [28]. The SEY properties for surface 256 materials can be defined by the following parameters: the 257 primary electron impact kinetic energies which yield $\delta = 1$, 258 W_1 , and W_2 ; the impact energy W_{max} necessary for a primary 259 electron to yield $\delta = \delta_{max}$, which is the maximum value of the 260 SEY function; and the value of the primary electron impact 261 energy $W_0(\delta = 0)$ that limits the region of elastic collisions. 262

When a multipactor discharge evolves in the partially dielectric-loaded waveguide under study, the dc field distribution has to be updated after each electron impacts on the 265

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Material	W _{max} (eV)	W ₁ (eV)	W ₂ (eV)	W ₀ (eV)	δ_{max}
Niobium	200.0	33	1500	-	1.6
Silver	165.0	30	5000	15.99	2.22
Teflon	271.7	27	5000	6.81	2.47

TABLE I Secondary Electron Emission Yield Properties of Different Materials [31], [32]

dielectric surface. However, tracking the evolution of all the 266 electrons involved in the multipactor discharge would suppose 267 a big computational cost. Thus, we have made use of an EEM, 268 but considering both the spread in secondary emission energy 269 and the angle of the secondary electrons after each impact on 270 the waveguide walls. This assumption has proved to account 271 properly for the charging of the dielectric material, given that 272 the discharging time for dielectrics is much higher than the 273 typical time for a multipactor discharge. Thus, in the EEM 274 assumed in this paper, after the effective electron impacts at 275 time t with any surface, $N_i(t)$ is modified according to the δ 276 value provided by the SEY function as follows: 277

$$N_i(t + \Delta t) = \delta N_i(t) \tag{16}$$

where $N_i(t)$ represents the population of the ee inside the waveguide at the instant *t*, and Δt is the time step used in the simulations.

The secondary electron departure kinetic energy E_s after each electron impact is assumed to fit the following probability density function [29]:

$$\frac{dp(E_s)}{dE_s} = C \exp\left[-\frac{\ln^2(E_s/E_m)}{2\tau^2}\right]$$
(17)

where *C* is a normalization constant, the parameter τ (typical values 0.7–0.8) determines the width of the distribution and *E_m* (typical values 3–4 eV) is the energy of the maximum of the spectrum. Finally, the secondary electrons after inelastic impacts are emitted following a cosine distribution of the polar angle.

III. NUMERICAL RESULTS AND DISCUSSION

An in-house simulation computer-aided design (CAD) tool 293 based on the Monte Carlo method described in Section II has 294 been developed to analyze the multipactor effect in partially 295 dielectric-loaded rectangular waveguides. The first problem 296 analyzed consists of an empty rectangular waveguide previ-297 ously studied in [14], whose multipactor prediction results 298 have been used for validation purposes. The rectangular 299 waveguide has dimensions a = 43.2 cm and b = 10.2 cm, 300 and is excited by a time-harmonic signal at f = 500 MHz. 301 The material of the waveguide walls is niobium, whose SEY 302 properties are given in Table I and can be expressed with 303 the simple model proposed in [30]. In the algorithm of the 304 simulator used in [14], for each RF power considered in the 305 waveguide, the initial electron is launched at x = a/2, and 306 the simulation is run 42 times, corresponding to 42 equidistant 307 phases of the RF field. The mean value of the final population 308 of electrons after 20 impacts of the ee against the walls is 309 calculated using all the 42 simulations. Also, the secondary 310



Fig. 2. Comparison with [14] of the mean value of N over all launch phases in a rectangular waveguide (a = 43.1 cm and b = 10.2 cm) driven at f = 500 MHz with a maximum of 20 impacts from a single initial launch location on the midline of the empty rectangular waveguide.

electrons generated after every collision are launched with an 311 energy of 2 eV normal to the impacting surface. The maximum 312 simulation lifetime of each ee is $t_{max} = 1000$ RF cycles, and 313 the simulation is stopped if the impact energy is lower than 314 0.1 eV or if the accumulated population of electrons is under 315 10^{-3} . To model the same simulation conditions, our CAD tool 316 has been adapted accordingly. In Fig. 2, the results of the mean 317 population of electrons, N, computed with our code (black 318 lines) are compared with the curves presented in [14] (gray 319 lines). In this figure, we can see some high-risk multipactor 320 power regions. Both curves show a good agreement in the 321 shape and location of these multipactor windows. 322

Once the model has been validated for an empty 323 waveguide, next we analyze the multipactor effect in a par-324 tially dielectric-loaded rectangular waveguide. The selected 325 waveguide configuration for the multipactor analysis is a 326 nonstandard silver-plated rectangular waveguide of width a =327 19.05 mm and heigh b = 0.4 mm, in which a thin dielec-328 tric layer has been placed over the bottom surface of the 329 waveguide. A realistic dielectric material has been chosen 330 as teflon (DuPont Teflon fluorinated ethylene propylene Flu-331 oroplastic Film Type), which is a dielectric film commonly 332 used in space applications, of thickness h = 0.025 mm and 333 $\epsilon_r = 2.1$; thus d = b - h = 0.375 mm. Standard values for 334 the SEY parameters of silver [31] are given in Table I, and 335 SEY parameters of teflon have been measured at the ESA-VSC 336 High Power Space Materials Laboratory (Valencia, Spain) 337 [32]. First, a study of the susceptibility chart of this waveguide 338 has been performed. Since it is a partially dielectric-loaded 339 rectangular waveguide, the factor $f \times d$ is plotted in the 340 horizontal axis of the susceptibility chart. In the vertical 341 axis, it is plotted an effective voltage, Veff, which has been 342 calculated numerically as the line integral of the E_{y} component 343 of the electric field (evaluated at the center of the waveguide 344 x = a/2 from $y_1 = 0$ to $y_2 = d$. To obtain this susceptibility 345 chart for each V_{eff} and $f \times d$ pair, the simulation is run 346 72 times, corresponding to 72 equidistant initial phases of 347 the RF field separated 5°. In each run, an initial single ee is 348 launched at x = a/2 and z = 0 and at a random position 349 y_0 in the y-axis between y = 0 and y = d. The initial 350 electron is launched with a departure kinetic energy given by 351 the probability density function shown in (17) and following 352



Fig. 3. Comparison of the susceptibility chart of a rectangular waveguide partially filled with teflon (black points) with that of its equivalent empty waveguide (gray points). Red: operating point corresponding to $f \times d = 3.13 \text{ GHz} \cdot \text{mm}$ and $V_{\text{eff}} = 608 \text{ V}$.

a cosine distribution of the polar angle. Each simulation was 353 stopped after 100 RF cycles. In the empty waveguide, the 354 arithmetic mean of the final population of electrons after 355 100 RF cycles is calculated using all the 72 simulations. 356 If this mean value is greater than 1, then the multipactor 357 discharge is assumed to have occurred. However, in a partially 358 dielectric-loaded waveguide, it has been shown in previous 359 works [8]–[10] that the emission or absorption of electrons 360 by the dielectric surface gives rise to an increasing dc field 361 in the waveguide, which eventually turns off the discharge. 362 Thus, in this case, a minimum mean value of the magnitude 363 of E_{dc} field in the waveguide after 100 RF cycles is used as the 364 criterion to assume that a multipactor discharge has occurred 365 at a given operating point. 366

Fig. 3 shows the computed susceptibility chart of the rec-367 tangular waveguide partially filled with teflon (black points). 368 The lowest $f \times d$ value is above the cutoff frequency 369 of the fundamental mode in this waveguide. In this figure, 370 the susceptibility chart of the equivalent empty waveguide 371 with the same vertical air gap is also represented with gray 372 points for comparison. It can be checked that both the empty 373 and the partially dielectric-loaded waveguide with the same 374 vertical air gap show similar multipactor susceptibility charts, 375 given that the SEY properties of silver and teflon are similar. 376 This susceptibility chart is not generally applicable to any 377 rectangular waveguide with an air gap d, given that the 378 electromagnetic field distribution depends on the geometry 379 and dimensions of the dielectric layer with respect to the 380 waveguide dimensions, and also on its relative permittivity. 381

From the results shown earlier, and with the purpose of 382 having a better understanding of the dynamics of the electron 383 inside the partially dielectric-loaded waveguide, a point within 384 the multipactor region has been chosen (highlighted in red 385 in Fig. 3), corresponding to $V_{\rm eff}$ = 608 V and $f \times d$ = 386 $3.13 \text{ GHz} \cdot \text{mm}$. In this case, the evolution of the multipactor 387 discharge in the partially dielectric-loaded waveguide under 388 study has been analyzed as a function of the time normalized 389 to the RF period. For this simulation, the electron is launched 390 with an initial phase of the RF field $\varphi_0 = 0^\circ$. Simulations 391 assuming different initial phases have been performed, and 392 similar results were obtained. The obtained simulation is 393 shown in Fig. 4, where it is plotted the y-coordinate followed 394 by the ee within the waveguide as a function of the normalized 395



Fig. 4. Trajectory (y-coordinate) of the ee in the air gap of the rectangular waveguide partially filled with teflon as a function of the RF cycle.



Fig. 5. Black line; time evolution of the total number of electrons N. Blue line; $E_{y,RF}$. Red line: $E_{y,dc}$ at the electron position.



Fig. 6. Distribution of normalized charges $\overline{Q}_i = Q_i/e$ appearing on the dielectric surface.

time. In the selected multipactor regime, which is inside the 396 multipactor region, the electron initially collides with the 397 top metallic and bottom dielectric surface consecutively in 398 what seems to be a first-order multipactor process during 399 the first 17 RF cycles, remaining in the vicinity of x =400 a/2 and z = 0—given that the electron has nearly no 401 acceleration in such directions. As shown in Fig. 5, in the 402 first cycles, the total number of electrons N (black solid 403 line) follows an exponential growth. This progressive growing 404 of N makes that the number of charges appearing on the 405 dielectric surface increases, the number which is proportional 406 to the emitted or absorbed electrons in each impact, as seen 407 in Fig. 6 (positive charges are represented with red circles, 408 while negative charges are represented with blue circles; the 409 circles' size is proportional to the charge magnitude in log 410 scale). Such charges on the dielectric interface give rise to the 411



Fig. 7. Transverse distribution of the dc electric field in the proximity of the main charge point in the waveguide air region at RF cycle 26.2, at z = 0.45 mm.

appearance of an electrostatic field in the empty gap during 412 the time between impacts. Once the population of electrons 413 reaches a significant number ($N \approx 10^9$ in the conditions under 414 study), the y-component of the dc field, $E_{y,dc}$ [which has been 415 plotted in Fig. 5 with red line at the positions (x, y, z) where 416 the effective electron is located in the displayed instants in 417 this figure] becomes comparable to $E_{y,RF}$, and the effective 418 electron is unable to keep up with its previous multipactor 419 synchronization. From this moment on, the dc field makes that, 420 in some impacts, the electrons collide with the top metallic 421 or bottom dielectric surface much sooner or later than the 422 instants when the RF electric field changes its sign, which 423 implies low impact energy collisions so that electrons are 424 absorbed in such impacts. In collisions at the dielectric surface, 425 the absorption or emission of electrons yield the appearance 426 of growing charges on the dielectric layer, contributing to a 427 higher dc field acting on the waveguide. The distribution of 428 this high dc field in the proximity of the main charge point 429 in the waveguide air region is shown in Fig. 7 at RF cycle 430 26.2 in the plane z = 0.45 mm (corresponding to the z 431 position of the electron at this instant). The action of this field 432 may result in the appearance of a single-surface multipactor 433 regime in the dielectric surface [see the y position of the 434 electron in Fig. 4 (inset) from RF cycle 26.2], with successive 435 low impact energy collisions, which eventually leads to the 436 turning off of the discharge itself (as can be appreciated in 437 Fig. 5 from RF cycle 26.2 on). From this instant, the dc field 438 distribution in the waveguide remains nearly constant, given 439 that N drops very quickly. The final value of the y-component 440 of the dc field accounts for the balance between the emitted 441 and absorbed electrons by the dielectric surface in the whole 442 process. Then, although the final population of electrons after 443 RF cycle 100 is 0, the remaining high dc field in the waveguide 444 indicates that a multipactor discharge has taken place in the 445 waveguide in this simulation. It is worth mentioning that the 446 observed turning OFF of the discharge observed in the last 447 stages of the multipactor evolution in this waveguide has been 448 speeded up due to the use of the EEM, although this does not 449 change qualitatively the dynamics of the discharge under these 450 conditions. 451

IV. CONCLUSION

A study of the multipactor effect in a partially dielectricloaded rectangular waveguide has been carried out. In this

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paper, we have considered the RF electromagnetic fields 455 (obtained with a very efficient vectorial modal method) as 456 well as the dc field caused by the appearance of a charge 457 distribution in the dielectric layer. The solution of the elec-458 trostatic problem has required the use of different numer-459 ical integration techniques and interpolation methods. The 460 electron trajectory has been numerically solved by using a 461 Velocity Verlet algorithm, providing sufficient accuracy and 462 good efficiency. As a first example, the multipactor prediction 463 results of an empty rectangular waveguide have been obtained 464 for validation purposes. Second, the susceptibility chart of 465 a partially dielectric-loaded rectangular waveguide has been 466 computed, and the time evolution of a discharge in this 467 waveguide has been studied and discussed. The performed 468 simulations reveal that multipactor discharges in this type of 469 dielectric-loaded waveguides turn OFF by themselves due to 470 the electrostatic field associated with the dielectric surface 471 charges that evolve with the multipactor process. 472

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